UNIVERSITY COLLEGE LONDON

University of London

EXAMINATION FOR INTERNAL STUDENTS

For The Following Qualification:-

M.Sc.

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Boundary Layers

COURSE CODE	: MATHG302
DATE	: 12-MAY-06
ТІМЕ	: 14.30
TIME ALLOWED	: 2 Hours

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All questions may be attempted but only marks obtained on question 5 and the best three other solutions will count. Question 5 must be attempted. The use of an electronic calculator is not permitted in this examination.

For questions 2 to 5, assume two-dimensional incompressible laminar flow.

The standard dimensional boundary layer equations for a boundary layer in the neighbourhood of y=0 are

 $u_x + v_y = 0$, $u_t + uu_x + vu_y = U_t + UU_x + vu_{yy}$

Here u and v are velocity components in the x and y directions respectively, t is time, ν is the kinematic viscosity of the fluid, and U(x,t) is the external flow in the x direction at y=0. Subscripts denote partial derivatives.

The streamfunction ψ is defined such that $u=\psi_y$ and $v=-\psi_x$. For length scale L and velocity scale U_0 , the Reynolds number is $R=U_0L/\nu$.

1. A two point boundary value problem for the function h(x) is defined by the differential equation

$$\epsilon(1+3x)h_{xx} - h = -C$$

where ϵ is a small positive parameter, C is a constant, and the boundary conditions are

$$h(0) = 1$$
 , $h(1) = 2$

What property makes this an example of a singular perturbation problem?

- (a) For the case C=2, determine the location of a boundary layer and find the first terms of the outer and inner asymptotic expansions.
 Provide a sketch of the resulting leading order solution for h(x).
 Find the second term of the inner expansion.
- (b) For the case C=1, again find the location of the boundary layer and the first terms of the inner and outer expansions.

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- 2. What is a basic property of flow represented by a similarity solution?
 - (a) Consider the standard boundary layer equations for steady flow in the case $U(x) = Cx^m$, where C > 0 and m are constants. By choosing a similarity variable $\eta = y/(Ax^n)$, where A and n are other constants, and a streamfunction of the form

$$\psi(x,y) = ACx^{m+n}f(\eta)$$

show that a similarity solution can be found if 2n = 1 - m. (Note that m and n need not be integers.)

By choosing $A^2 = \nu/[C(m+n)]$, derive the Falkner-Skan equation

$$f''' + ff'' + m(1 - f'^2)/(m + n) = 0 .$$

When $C = U_0/L^m$ (i.e. $U(x) = U_0(x/L)^m$), show that the similarity variable can be written as

$$\eta = R^{1/2}(m+n)^{1/2}(y/L)(x/L)^{-n}$$
.

(b) The streamfunction for a steady inviscid flow in polar co-ordinates is

$$\psi = (U_o L/\lambda)(r/L)^{\lambda} \sin[\lambda(\theta - \pi)] ,$$

where $\lambda = 3/2$.

Given that $u^{(r)} = \psi_{\theta}/r$ and $u^{(\theta)} = -\psi_r$, find expressions for these velocity components.

Show that the lines $\theta = \pi/3$, $\theta = \pi$ and $\theta = 5\pi/3$ are streamlines. Sketch the streamlines in the region $\pi/3 \le \theta \le 5\pi/3$.

(c) How might the result in (a) be used to analyse the boundary layers for flow past a wedge, for which the inviscid flow is that given in (b)? What would be appropriate values for m and n in this case?

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- 3. Consider steady flow past a flat plate that lies along y=0, between x=-L and x=0. The flow is symmetric about y=0, and upstream of the plate the flow is $(u, v)=(U_0, 0)$ where U_0 is a positive constant.
 - (a) In the wake downstream of the plate the standard boundary layer equations can be used. From those equations, prove that the quantity

$$\int_0^\infty u(U_0-u) \, dy$$

is independent of x.

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(b) Far downstream, suppose $u/U_0 = 1 - F(x, y)$, where $F \ll 1$. Derive the linearised boundary layer equation

$$F_x = A^2 F_{yy} ,$$

where $A^2 = \nu/U_0$.

(c) By using the substitution $F=f(\eta)/x^{1/2}$, where $\eta=y/(Ax^{1/2})$, derive the ordinary differential equation

$$f'' + (\eta f' + f)/2 = 0$$

Justify the boundary conditions f'=0 at $\eta=0$ and $f \to 0$ as $\eta \to \infty$. Find f to within an arbitrary multiplicative constant. How might the value of that constant be determined?

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4. Suppose flow past an obstacle is impulsively started from rest at time t=0, such that the inviscid flow at the surface of the obstacle, along y=0, is given to be U(x) for t > 0.

Given that the streamfunction for the flow in the boundary layer has the form

$$\psi = 2(\nu t)^{1/2} [UF_0(\eta) + tUU_xF_1(\eta) + \text{terms of order } t^2]$$
,

where $\eta = y/(4\nu t)^{1/2}$, find corresponding expressions for u and v.

(a) Derive the ordinary differential equation

$$F_0''' + 2\eta F_0'' = 0$$

for F_0 , and also derive three boundary conditions.

(b) Given that

$$\int_0^\infty e^{-\lambda^2} d\lambda = \sqrt{\pi}/2 \quad ,$$

prove that to leading order

$$u = U(x)(2/\sqrt{\pi})\int_0^{\eta} e^{-\lambda^2} d\lambda$$

(c) For the case $U(x) = U_0(1 + e^{-x^2/L^2})$, find the value of x where separation is first expected to occur on the surface of the obstacle. Hence show that separation is first expected at time

$$t = (L/U_0) (2e/\pi)^{1/2} / F_1''(0)$$

(You may assume that $F_1''(0)$ is positive.)

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5. With velocity scaled by U_0 , distance scaled by L, and pressure scaled by ρU_0^2 , dimensionless equations for steady flow are

$$uu_x + vu_y = -p_x + R^{-1}(u_{xx} + u_{yy}) \quad , \tag{1}$$

$$uv_x + vv_y = -p_y + R^{-1}(v_{xx} + v_{yy}) \quad , \tag{2}$$

with $R^{-1} \ll 1$, and $u_x + v_y = 0$. Consider flow past a flat plate lying along y=0, between x=-1 and x=0, with (u,v)=(1,0) far from the plate.

In the context of matched asymptotic expansions, why is a boundary layer required near the plate?

If the inner variable is $Y=y/\epsilon$, state (without proof, but giving a reason) the choice of ϵ required to obtain the standard boundary layer equations.

(a) For the triple deck theory required near the trailing edge of the plate, appropriate scales for the upper deck are $W=y/\delta^3$ and $X=x/\delta^3$, with expansions of the form

$$\psi \sim y - \delta^4 C + \delta^5 F(X, W)$$

 $p \sim \delta^2 P(X, W)$

where $\delta^4 = R^{-1}$ and C is a constant.

What does the term $-\delta^4 C$ represent?

With these scalings, rewrite (1) and (2) in terms of F and P.

Deduce that $F_{WX} = -P_X$ and $F_{XX} = P_W$.

Given $V(X, W) = -F_X = f(X)g(W)$, show that

$$V_{XX} + V_{WW} = f_{XX}g + fg_{WW} = 0$$

(b) The boundary conditions for V are $V \to 0$ as $X \to \pm \infty$ and as $W \to \infty$, and $V \to -A_X$ as $W \to 0$ for some function A(X). Prove that

$$V = -(2\pi)^{-1} \int_{-\infty}^{\infty} A_{\lambda} \frac{2W}{W^2 + (X-\lambda)^2} d\lambda \quad .$$

You are given that

$$\int_{-\infty}^{\infty} e^{ik(X-\lambda)} e^{-|k|W} dk = \frac{2W}{W^2 + (X-\lambda)^2}$$

and you may use the Fourier transform relations

$$\hat{f}(k) = (2\pi)^{-1/2} \int_{-\infty}^{\infty} f(X) e^{-ikX} dX , \quad f(X) = (2\pi)^{-1/2} \int_{-\infty}^{\infty} \hat{f}(k) e^{ikX} dk .$$

(c) Given that $P \to 0$ as $W \to \infty$, and that $V_X = -P_W$, use the above expression for V to find a similar expression for P.

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