

1. For the singular perturbation problem

$$\epsilon y'' + y' + y^2 = 0 \quad (0 < \epsilon \ll 1), \quad y(0) = 2, \quad y(1) = \frac{1}{2},$$

seek an outer expansion

$$y(x) = y_0(x) + \epsilon y_1(x) + \dots$$

Explain briefly why the inner (boundary) layer must be at  $x = 0$ , and hence determine the functions  $y_0(x)$  and  $y_1(x)$ .

Seek an inner expansion

$$y(x) = Y_0(X) + \epsilon Y_1(X) + \dots,$$

and calculate the first two terms, where  $X = x/\epsilon$ .

Use the two-term outer and inner expansions to construct a composite solution  $C_{11}y$  based on the additive rule.

2. A three-dimensional flow past a flat plate of length  $L$  is described in a coordinate system  $(x, y, z)$  normalised by  $L$ , where  $x$  and  $z$  are on the plate surface with  $x$  and  $z$  being parallel and perpendicular to the leading edge respectively, while  $y$  is normal to the plate. The corresponding velocity components, normalised by a reference velocity  $U_\infty$ , are denoted by  $(u, v, w)$ , and the normalised pressure is denoted by  $p$ .

Assuming that the Reynolds number  $R = U_\infty L / \nu \gg 1$ , where  $\nu$  is the kinematic viscosity, write down the governing equations for the inviscid approximation  $(u_0, v_0, w_0, p_0)$ , and specify appropriate boundary conditions.

By a brief scaling argument, deduce that the boundary layer lies in a layer  $y \sim O(R^{-1/2})$ , and that the expansion for  $(u, v, w, p)$  in the boundary layer takes the form

$$(u, v, w, p) = \left( U(x, Y, z), R^{-1/2} V(x, Y, z), W(x, Y, z), P(x, z) \right) + \dots,$$

where  $Y = R^{1/2} y$ . Derive the boundary-layer equations, specify the appropriate boundary and matching conditions, and determine the pressure gradients  $P_x$  and  $P_z$  in terms of relevant slip velocities.

If the inviscid slip velocities are

$$u_0 = x^m, \quad w_0 = \beta \text{ (constant)},$$

there exists a similarity solution of the form

$$U = x^m f'(\eta), \quad V = -\frac{1}{2} x^{(m-1)/2} \left[ (m+1)f + (m-1)\eta f'(\eta) \right], \quad W = g(\eta),$$

where  $\eta = Y/s(x)$ .

Deduce that  $s(x) = x^{(1-m)/2}$ , and show that  $f$  and  $g$  satisfy the equations

$$f''' + \frac{1}{2}(m+1)ff'' + m(1-f'^2) = 0, \quad g'' + \frac{1}{2}(m+1)fg' = 0,$$

and state the boundary and matching conditions.

For the special case  $m = 0$ , deduce a simple relation between  $f$  and  $g$ .

3. In a suitably non-dimensionalised coordinate system  $(x, y)$ , a steady flow is described by the Navier-Stokes equations

$$\left. \begin{aligned} u_x + v_y &= 0 \\ uu_x + vv_y &= -p_x + R^{-1}(u_{xx} + u_{yy}) \\ uv_x + vv_y &= -p_y + R^{-1}(v_{xx} + v_{yy}) \end{aligned} \right\} .$$

Suppose that the surface pressure gradient is such that in the vicinity of a point  $x_s$ , the boundary layer velocity field,  $(U_0(x, Y), R^{-\frac{1}{2}}V_0(x, Y))$ , has the property that

$$U_0(x, Y) \rightarrow Y^2 \quad \text{as } Y \rightarrow 0; \quad U_0(x, Y) \rightarrow 1 \quad \text{as } Y \rightarrow \infty,$$

where  $Y = R^{\frac{1}{2}}y$ .

When the flow is perturbed by a sudden localised perturbation in the region  $x - x_s = O(D) \ll 1$ , the solution in the main part of the boundary layer can be sought in the form

$$u = U_0 + \epsilon U_1(X, Y) + \dots, \quad v = R^{-\frac{1}{2}}V_0 + \frac{\epsilon R^{-\frac{1}{2}}}{D}V_1(X, Y) + \dots,$$

where  $\epsilon \ll 1$ , and  $X = (x - x_s)/D$ .

Derive the equations for  $U_1$  and  $V_1$ , and find the solution for  $U_1$  and  $V_1$ .

Examine the behaviour of  $V_1$  as  $Y \rightarrow \infty$ , and explain why an upper deck is needed. Estimate the order of magnitude of the pressure in the upper deck.

Examine the behaviour of  $U_1$  as  $Y \rightarrow 0$ , and deduce the width of this layer in terms of  $D$  and  $R$ . Estimate the streamwise velocity of the perturbation and the inertia in terms of  $D$  and  $\epsilon$ .

Show that the flow becomes interactive when

$$D = O(R^{-2/7}).$$

Comment briefly on the role of each of the three decks.

Assuming that  $\epsilon$  is sufficiently small, deduce the expansion for the flow field in the lower-deck and derive the governing equations, and specify the matching condition with the main-deck solution.

4. The boundary-layer flow of a stratified fluid subject to a localised unsteady suction may be described by triple deck theory, in which the lower deck is governed by the equations

$$\left. \begin{aligned} u_X + v_y &= 0, & u_t + uu_X + vv_y &= -P_X + u_{yy} \\ u = 0, \quad v = v_s(X, t) & \text{ on } y = 0 \\ u \rightarrow y + A(X, t) & \text{ as } y \rightarrow \infty \end{aligned} \right\}.$$

The pressure  $P$  is related to the upper deck pressure  $p$  via

$$P = p(X, 0, t).$$

In the upper deck, the pressure  $p(X, \bar{y}, t)$  satisfies

$$\frac{\partial^2 p}{\partial X^2} + \frac{\partial^2 p}{\partial \bar{y}^2} + \gamma^2 p = 0,$$

subject to the boundary condition

$$\frac{\partial p}{\partial \bar{y}} = A_{XX} + \gamma^2 A \quad \text{on } \bar{y} = 0; \quad p \text{ is finite as } \bar{y} \rightarrow \infty,$$

where  $\gamma > 0$  is a constant representing the effect of stratification.

For a weak suction  $v_s = HV_s(X) e^{-i\omega t} + c.c.$  with  $H \ll 1$ , seek a solution of the form

$$(u, v, P, A) = (y, 0, 0, 0) + H(u_1(y), v_1(y), P_1, A_1) e^{-i\omega t} + O(H^2).$$

Derive the linearised system including the boundary and matching conditions, and show that

$$-i\omega u_{1,y} + y u_{1,Xy} - u_{1,yyy} = 0, \quad u_{1,yy}(X, 0) = P_{1,X} + V_s(X).$$

Suppose that  $\widehat{P}_1$  is the Fourier transform of  $P_1$ , i.e.

$$\widehat{P}_1 = \int_{-\infty}^{\infty} P_1(X) e^{-ikX} dX.$$

Solve the system by Fourier transform to show that

$$\widehat{P}_1 = -\frac{\widehat{V}_s(k)}{ik\Delta(k, \omega)} \int_{\zeta_0}^{\infty} Ai(\zeta) d\zeta \quad \text{with } \zeta = (ik)^{\frac{1}{3}}y + \zeta_0, \quad \zeta_0 = -i\omega(ik)^{-2/3}$$

where  $\widehat{V}_s$  is the Fourier transform of  $V_s$ , and

$$\Delta(k, \omega) = \int_{\zeta_0}^{\infty} Ai(\zeta) d\zeta + i(ik)^{2/3} Ai'(\zeta_0) / (k(k^2 - \gamma^2)^{1/2}),$$

with  $Ai$  being Airy function.

5. The stability of a two-dimensional boundary layer is studied by introducing a two-dimensional disturbance, and the perturbed flow field is written as

$$(u, v, p) = \left( U_0(x/R, y), R^{-1}V_0(x/R, y), 0 \right) + \epsilon(\tilde{u}, \tilde{v}, \tilde{p}),$$

where  $x, y$  are non-dimensionalised by a reference boundary layer thickness  $\delta$ , the Reynolds number  $R$  is based on  $\delta$ , and  $\epsilon \ll 1$  represents the magnitude of the perturbation.

Given that  $(u, v, p)$  satisfy the Navier-Stokes equations

$$\left. \begin{aligned} u_x + v_y &= 0 \\ u_t + uu_x + vu_y &= -p_x + \frac{1}{R}(u_{xx} + u_{yy}) \\ v_t + uv_x + vv_y &= -p_y + \frac{1}{R}(v_{xx} + v_{yy}) \end{aligned} \right\},$$

derive the linearised equations governing the perturbation  $(\tilde{u}, \tilde{v}, \tilde{p})$ . Indicate the terms which represent the non-parallel-flow effect.

Explain what is meant by Prandtl's parallel-flow approximation.

Suppose that this approximation is employed to seek a normal-mode solution of the form

$$(\tilde{u}, \tilde{v}, \tilde{p}) = \left( \hat{u}(y), \hat{v}(y), \hat{p}(y) \right) e^{i\{\alpha x - \omega t\}} + c.c..$$

Derive the equations governing  $\hat{u}, \hat{v}, \hat{p}$ , and hence show that  $\hat{v}$  satisfies

$$\left\{ (U_0 - \omega/\alpha) \left( \frac{\partial^2}{\partial y^2} - \alpha^2 \right) - U_{0,yy} - (i\alpha R)^{-1} \left( \frac{\partial^2}{\partial y^2} - \alpha^2 \right)^2 \right\} \hat{v} = 0.$$

Comment on the appropriateness of this approach for explaining boundary-layer instability, in the following two cases:

- (1)  $U_0$  has an inflection point;
- (2)  $U_0$  has no inflection point.

In the case (2) above, relevant normal modes are of long wavelength in the sense that

$$\alpha \sim O(R^{-1/4}), \quad \omega \sim O(R^{-1/2}).$$

Show that viscosity is negligible for  $y \sim O(1)$ . Assuming that  $U_0 \sim y$  as  $y \rightarrow 0$ , deduce that viscosity is a leading-order effect in a sublayer where  $y \sim O(R^{-1/4})$ . Show that in the  $x$ -momentum equation for the perturbation, the ratio of the non-parallel-flow term to the viscous term is  $O(R^{-3/4})$ .